

THEORETICAL AND PRACTICAL ASPECTS FOR THE STRESS STATE

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ABSTRACT: *In the present study we propose to determine the knowing the state of stress at a point in a deformable geometrical solid involves knowing the unitary stresses acting on three surfaces (defined in three planes) perpendicular to each other that pass through the point considered. On each surface there acts a normal stress and a tangential stress, represented by the two components parallel to the common edges. Due to the duality of tangential tensions, they are equal two by two, so the stress state at one point is defined by six stress, three normal and three tangential. Applying the formulas corresponding to the state of tension for a given deformable geometrical solid to which known $\sigma_x, \sigma_y, \tau_{xy}$ determine the main stresses and the main directions that characterize the state of deformation.*

KEY WORDS: deformable geometrical solid, normal stress, tangential stress, main stresses, main directions.

1. INTRODUCTION

At a point in a deformable geometrical solid, the state of tension can be defined by the tensions acting on three surfaces (defined in three planes) perpendicular between them, passing through the point considered. On each surface there acts a normal stress and a tangential stress, represented by the two components parallel to the common edges [1]. Knowing the state of stresses and deformations in a required material involves knowing the unitary stresses and the specific deformations at each point of the material, as well as the relationships between stresses and specific deformations around a point.

The stress state at any point in the deformable solid is perfectly determined if the stresses are known on three coordinate planes that pass through that point [2]. If the position of the coordinate planes is arbitrary, each of these planes has, in the general case, both normal stresses and tangential stresses.

2. THE STATE OF TENSION

It is considered a plate of constant thickness requested by a system of forces in equilibrium. All forces act in the median plane of the plate. In this situation the stresses appearing at any point in the body are parallel to the plane of the plate, and the plane requirement is identical in all planes parallel to the median plane (3).

This request produces what is called the flat tension state. If the plate were fixed between two rigid walls, in order to prevent transverse deformations, so that the deformations occur only in planes parallel to the median plane, there could be a flat state of deformation. Using three planes perpendicular to the plane of the plate, a triangular prism is isolated with the base of the OBC rectangular triangle OBC (fig.1).

Using a xOy rectangular axis system, normal stresses σ , will bear the axis index with which they are parallel, and the tangential stresses τ will have two indices, the first is the direction of their own direction, and the second the normal direction index to the plane in which they act (3).

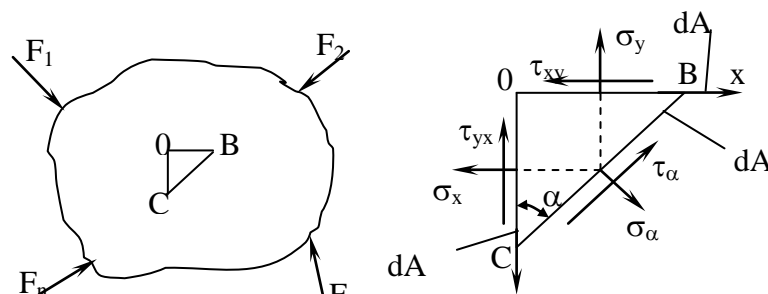


Fig.1. The state of tension

On the BC plane, making the angle α with the vertical plane OC, both tensions will carry the α index.

Writing with dA the surface area of the inclined surface BC of the element, the OB and OC surface areas will be $dA \sin \alpha$ and $dA \cos \alpha$.

The detached element on the plate will be in balance under the action of the forces that act as the plate. It is written a first equation of equilibrium as the sum of moments calculated from the point D, where the resultant of the three tensions σ intersect:

$$\sum M_D = \tau_{xy} dA \sin \alpha \cdot \frac{BC}{2} \cos \alpha - \tau_{yx} dA \cos \alpha \cdot \frac{BC}{2} \sin \alpha = 0$$

Result: $\tau_{xy} = \tau_{yx}$, the law of parity or duality of tangential tensions.

Two more equations of equilibrium are written, projecting the forces in the direction of the stresses σ_α and τ_α (3):

$$\sigma_\alpha dA - \sigma_x dA \cos \alpha \cos \alpha - \sigma_y dA \sin \alpha \sin \alpha - \tau_{xy} dA \sin \alpha \cos \alpha - \tau_{yx} dA \cos \alpha \sin \alpha = 0$$

$$\tau_\alpha dA - \sigma_x dA \cos \alpha \sin \alpha + \sigma_y dA \sin \alpha \cos \alpha - \tau_{xy} dA \sin \alpha \sin \alpha + \tau_{yx} dA \cos \alpha \cos \alpha = 0$$

Taking into account the law of parity ($\tau_{xy} = \tau_{yx}$) in the equations obtained, we can express the stresses σ_α and τ_α on the inclined face of the element, depending on the known stresses, on the perpendicular faces and according to the angle of obliquity α :

$$\sigma_\alpha = \sigma_x \cos \alpha + \sigma_y \sin^2 \alpha + 2\tau_{xy} \sin \alpha \cos \alpha; \tau_\alpha = (\sigma_x - \sigma_y) \sin \alpha \cos \alpha - \tau_{xy} (\cos^2 \alpha - \sin^2 \alpha)$$

Entering the double angle will result:

$$\sigma_\alpha = \frac{\sigma_x + \sigma_y}{2} + \frac{\sigma_x - \sigma_y}{2} \cos 2\alpha + \tau_{xy} \sin 2\alpha \quad (1)$$

$$\tau_\alpha = \frac{\sigma_x - \sigma_y}{2} \sin 2\alpha - \tau_{xy} \cos 2\alpha$$

The stresses σ_α and τ_α , as a function of the angle α (1), can take maximum or minimum values for certain α angle values. Such angular values define the so-called main directions of tension. Maximum and minimum stresses are called the main stresses.

The main directions are obtained mathematically by canceling the derivatives of relations (1):

$$\frac{d\sigma}{d(2\alpha)} = -\frac{\sigma_x - \sigma_y}{2} \sin 2\alpha + \tau_{xy} \cos 2\alpha = 0 \quad (2)$$

$$\frac{d\tau}{d(2\alpha)} = \frac{\sigma_x - \sigma_y}{2} \cos 2\alpha + \tau_{xy} \sin 2\alpha = 0 \quad (3)$$

Relationship (2) will give:

$$\operatorname{tg} 2\alpha_{1,2} = \frac{2\tau_{xy}}{\tau_x - \tau_y} \quad (4)$$

and the relationship (3):

$$\operatorname{tg} 2\alpha_{3,4} = -\frac{\sigma_x - \sigma_y}{2\tau_{xy}} \quad (5)$$

Equations (4) and (5) have two solutions offset with 180° :

$$2\alpha_2 = 2\alpha_1 + \pi; \quad \alpha_2 = \alpha_1 + \frac{\pi}{2}; \quad (6)$$

$$2\alpha_4 = 2\alpha_3 + \pi; \quad \alpha_4 = \alpha_3 + \frac{\pi}{2} \quad (7)$$

There are two main directions for each of the voltages σ and τ , the directions forming between them a right angle. After a main direction the voltage is the maximum value, and after the second the minimum value.

From the comparison of relations (1) and (2) we can deduce:

$$\frac{d\sigma}{d(2\alpha)} = -\tau = 0 \quad (8)$$

which shows that in the planes where normal tensions are the main ones, tangential tensions are missing.

From the comparison of relations (4) and (5) we deduce that, $\operatorname{tg} 2\alpha_{1,2} \operatorname{tg} 2\alpha_{3,4} = -1$

$$\text{so, } 2\alpha_{3,4} = 2\alpha_{1,2} + \frac{\pi}{2} \quad \text{or} \quad 2\alpha_{3,4} = 2\alpha_{1,2} + \frac{\pi}{4} \quad (9)$$

for example, the planes in which the tangential stresses are maximum or minimum, make angles of 45° with the planes in which the normal stresses are the main ones.

By replacing the angles α given by the relations (4) and (5) in relations (1) the main stresses will result in the form (3),

$$\sigma_{1,2} = \frac{\sigma_x + \sigma_y}{2} \pm \sqrt{\left(\frac{\sigma_x - \sigma_y}{2}\right)^2 + \tau_{xy}^2} \quad (10)$$

$$\sigma_{3,4} = \pm \sqrt{\left(\frac{\sigma_x - \sigma_y}{2}\right)^2 + \tau_{xy}^2} \quad (11)$$

Relationships (10) will, $\sigma_1 + \sigma_2 = \sigma_x + \sigma_y = \text{constant}$, (12)

which shows that the sum of the normal tensions from two perpendicular planes, taken at one and the same point, does not depend on the angle α , is invariant.

According to relations (11) the maximum tangential tension is equal to the minimum. This is in accordance with the law of duality of tangential tensions.

Based on the relationship (10) it can be written,

$$\tau_{\max} = \frac{1}{2}(\sigma_1 - \sigma_2) \quad (13)$$

this means that the maximum tangential voltage is the half-dimension of the main normal stresses.

In fig. 2 are the main normal stresses and their main directions around an arbitrary point. Considering a element in the vicinity of the contour, which is assumed to be unloaded, the element's faces will lack the tangential stresses. As a result, the main directions of contour stresses will be normal and tangential to contour respectively (3).

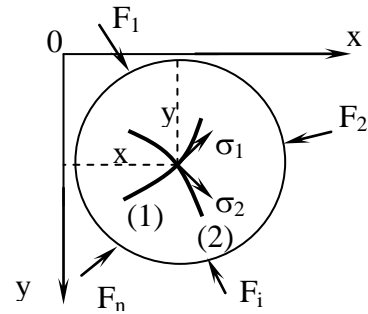
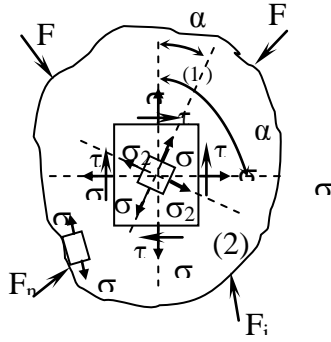


Fig.2. Main normal stresses and main directions Fig.3. Maximum and Minimum Requests Direction

For a flat state of tension, a network of tangent lines can be drawn at all their points at the main stress tensions. These are called isostatic curves and they indicate the direction of maximum and minimum stresses (fig. 3).

The equation of the isostatic lines is easily obtained by the relationship (4):

$$\operatorname{tg} 2\alpha_{1,2} = \frac{2\operatorname{tg} \alpha_{1,2}}{1 - \operatorname{tg}^2 \alpha_{1,2}} = \frac{2y'_{1,2}}{1 - y'^2_{1,2}} = \frac{2\tau_{xy}}{\sigma_x - \sigma_y} \quad (14)$$

from which it follows,

$$y'^2_{1,2} + \frac{\sigma_y - \sigma_x}{\tau_{xy}} y'_{1,2} - 1 = 0 \quad (15)$$

and solutions,

$$y'^2_{1,2} = \frac{\sigma_x - \sigma_y}{2\tau_{xy}} \pm \sqrt{\left(\frac{\sigma_y - \sigma_x}{2\tau_{xy}}\right)^2 + 1} \quad (16)$$

which represents two families of perpendicular curves, because, $y'_1 \cdot y'_2 = -1$

Tangential curves to maximum tangential stresses are called sliding lines. The differential equation of these lines will be obtained as before, in the next form (3):

$$y'_{3,4} = \frac{2\tau_{xy}}{\sigma_x - \sigma_y} \pm \sqrt{\left(\frac{\sigma_x - \sigma_y}{2\tau_{xy}}\right)^2 + 1} \quad (17)$$

Relationships (1) allow the expression of the stresses σ_α and α from any section at any one point, depending on the main stresses.

For this you must as: $\sigma_x = \sigma_1$, $\sigma_y = \sigma_2$, $\tau_{xy} = 0$:

$$\sigma_\alpha = \frac{\sigma_1 + \sigma_2}{2} + \frac{\sigma_1 - \sigma_2}{2} \cos 2\alpha$$

$$\tau_\alpha = \frac{\sigma_1 - \sigma_2}{2} \sin 2\alpha$$
(18)

By eliminating the angle 2α ,

$$\left(\sigma_\alpha - \frac{\sigma_1 + \sigma_2}{2}\right)^2 + \tau_\alpha^2 = \left(\frac{\sigma_1 - \sigma_2}{2}\right)^2$$
(19)

3. PRACTICAL CONSIDERATIONS

This calculation can be applied to a parallelepiped shape volume element having known (4) (Fig. 3) $\sigma_x=80\text{MPa}$, $\sigma_y=40\text{MPa}$ și $\tau_{xy}=50\text{MPa}$

The formula (10) is used to calculate the main stresses:

$$\sigma_{1,2} = \frac{\sigma_x + \sigma_y}{2} \pm \sqrt{\left(\frac{\sigma_x - \sigma_y}{2}\right)^2 + \tau_{xy}^2} = \frac{80+40}{2} \pm \sqrt{\left(\frac{80-40}{2}\right)^2 + 50^2}$$

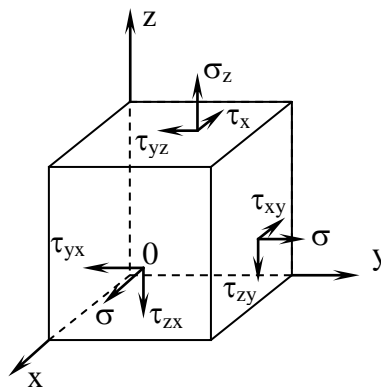


Fig. 4. Tension distribution

The result is $\sigma_1 = 113,851\text{MPa}$ and $\sigma_2 = 1,149\text{ MPa}$

The main directions are determined by the formula (14):

Then $\alpha_1 = 34,1^\circ$ and $\alpha_2 = 124,1^\circ$

3. CONCLUSIONS

For a plate form of constant thickness requested by a system of forces in equilibrium all forces act in the median plane of the plate.

The stresses appearing at any point in the body are parallel to the plane of the plate and the plaque requirement is identical in all planes parallel to the median plane. If the plate is fixed between two rigid walls there could be a flat state of deformation.

In this paper we did the calculation for a parallelepiped shape volume element and we determined the main direction and the main stresses that are oriented about α angle.

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